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Supersonic wind tunnel testing

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Forces and moments measurements by the 6-component external balance

- Measurement range: 2kN, 1.5kN, 200Nm, 50Nm
- External balance fixed in a disk-shaped housing mounted in a cavity in a side wall of the tunnel
- Model with a fixed pin mounted to the balance
- Angle of attack change provided by disk rotation
- Time series of 6-components (forces and moments) available







NACA 0012 airfoil

- Large oscilations of forces and moments
- Reversal of normal (lift) and axial forces
- Resonance frequency depending on the type and weight of the model

Diamond profile

- Good directions of forces
- Very limited range of angles of attack

Visualisations by Schlieren optical technique

- Schlieren optical system
- 270 mm diameter mirrors
- continous light source
- high speed camera acquisition (3000fps)



Wind tunnel choking

Cylindrical model- influence of diameter size



Regulated diffuser throat height



- improved characteristics, low noise
- wind tunnel opening/close peaks
- high angle of attack range
- allows for full $c_{L}(\alpha)$ characteristics

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Conclusions and perspectives

- preliminary tests of supersonic wind tunnel have been performed on NACA 0012 airfoil, diamond profile and other models
- forces and moments measurements has been done with use of an external balance as well as visualisation by Schlieren optical technique
- choking problem has been solved by regulating the diffuser opening height
- quantitative results of forces and moments has already been obtained for different models
- use of internal balance would allow for simultaneous visualisation and comparison of visible phenomena with acquired data